

LIFE LIMITED PARTS DATA SHEET

APU MODEL: GTCP36-150CL SERIAL NUMBER: P-292C

C&L Aerospace										
		APU								
PART NUMBER SERIAL NUMBER DATE OF OVERHAUL CURRENT ASSY TSN CURRENT ACFT TAT INSTALLED		ASSY TSN @ INSTALL	ACFT SN	ACFT REG	ACFT TAT @ INSTALL	CURRENT ASSY TSO				
P-292C	June-96 6742.0		6104.5	27-Mar-15	4697	5314	2-CLES	4617.8	2045.0	
LIFE LIMITED PARTS										
PART NUMBER	SERIAL NUMBER	LIFE LIMIT (FH)	DATE INSTALLED	PART TSN @ INSTALL	ASSY TSN @ INSTALL	CURRENT PART TSN	LIFE REMAINING			
3622632-1	14-162045-01841	20000	27-Mar-15	0	4697	2045.0	17955.0			
3842368-1	14-156101-04782	10000	27-Mar-15	0	4697	2045.0	7955.0			
		SB STATUS								
REV.	DESCRIPTION	ON	DATE C/W							
6	Standard storage and pres	ervation guidelines	30-Mar-22							
0	Workscope planning guide	(WPG 36-150 series)	8-Mar-15							
0	Conversion of GTCP36-100E to	36-150CL configuration	27-Mar-15							
1	Replacement of load cont	rol valve coupling	26-Mar-15							
2	Rotating group balanc	ing procedure	26-Mar-15							
	SERIAL NUMBER P-292C PART NUMBER 3622632-1 3842368-1 REV. 6 0 0 1	SERIAL NUMBER P-292C June-96 PART NUMBER SERIAL NUMBER 3622632-1 14-162045-01841 3842368-1 14-156101-04782 REV. DESCRIPTION 6 Standard storage and presonance of the presonance of t	SERIAL NUMBER DATE OF OVERHAUL CURRENT ASSY TSN P-292C June-96 CIFE LIMITED PAR' PART NUMBER SERIAL NUMBER LIFE LIMIT (FH) 3622632-1 14-162045-01841 20000 3842368-1 14-156101-04782 10000 SB STATUS REV. DESCRIPTION 6 Standard storage and preservation guidelines 0 Workscope planning guide (WPG 36-150 series) 0 Conversion of GTCP36-100E to 36-150CL configuration 1 Replacement of load control valve coupling	SERIAL NUMBER DATE OF OVERHAUL CURRENT ASSY TSN ACFT TAT P-292C June-96 6742.0 6104.5 LIFE LIMITED PARTS PART NUMBER SERIAL NUMBER LIFE LIMIT (FH) DATE (NSTALLED) 3622632-1 14-162045-01841 20000 27-Mar-15 3842368-1 14-156101-04782 10000 27-Mar-15 SB STATUS REV. DESCRIPTION DATE C/W 6 Standard storage and preservation guidelines 30-Mar-22 0 Workscope planning guide (WPG 36-150 series) 8-Mar-15 0 Conversion of GTCP36-100E to 36-150CL configuration 27-Mar-15 1 Replacement of load control valve coupling 26-Mar-15	SERIAL NUMBER DATE OF OVERHAUL CURRENT ASSY TSN CURRENT COURTENT COURTENT	SERIAL NUMBER DATE OF OVERHAUL CURRENT ASSY TSN CURRENT ACFT TAT INSTALLED INSTALLED INSTALLED INSTALLED	SERIAL NUMBER DATE OF OVERHAUL CURRENT ASSY TSN CURRENT COURTENT COURTENT	APU	SERIAL NUMBER DATE OF OVERHAUL CURRENT ASSYTSN CURRENT INSTALLED I	

1 of 1 4/11/2022

ol 2) Time in Air (Col. 3) Total Time Hrs. Mins. Hrs. Mins. Cycles APU MODE	(Col. 4) Enter Full details Mainten EL: GTCP36-150(CL) S/N: TSN: 6742.0 maintained under the manufacture	of Inspections, Minor Repairs and lance hereunder P-292C DATE: 13FEB2020 Page 1 of 1 Prer's inspection (MRB) as per CFR title with under this work order but not liste	(Col. 5) * Signature and Date				
1. This engine is currently being no 91.409(f)(3). 2. Reference CLAS W/O: A1137 fo	TSN: 6742.0 maintained under the manufacture	Page 1 of 1 arer's inspection (MRB) as per CFR title					
91.409(1)(3). 2. Reference CLAS W/O: A1137 fo							
		Har this work order but not liste	a on this				
CUSTOMER REQUESTED TASK							
 Preserve APU prior to removal – Preserved APU P/N: 3800804-1 S/N: P-292C IAW Honeywell 36-150(CL) MM 49-20-00 Rev 1, dated May 15/114 Removed APU per customer request – Removed APU P/N: 3800804-1 S/N: P-292C from CL604 S/N: 5314 at ACTT: 6104.50 ACTC: 4131. IAW CL604 AMM 49-10-004 							
I certify the above maintenance/inspection was performed in accordance with current Federal Aviation Regulations. All inspections were performed in accordance with 91.409(f)(3). Only the items specified in the Work Order have been found to be Airworthy and is Returned to Service. No other representation concerning Airworthiness are expressed or implied. Pertinent details of work performed on file at this repair station under work order no. A1137.							
	For: C&L	Aviation Services, LLC CRS#J2SR196J	4				
•	 Preserve APU prior to remove 49-20-00 Rev 1, dated May 15/4. Removed APU per customer ACTT: 6104.50 ACTC: 4131. IAV I certify the above maintenance/inspection we performed in accordance with 91.409(f)(3). Oto Service. No other representation concerning. 	 Preserve APU prior to removal – Preserved APU P/N: 3800804 49-20-00 Rev 1, dated May 15/114 Removed APU per customer request – Removed APU P/N: 38 ACTT: 6104.50 ACTC: 4131. IAW CL604 AMM 49-10-004 I certify the above maintenance/inspection was performed in accordance with curperformed in accordance with 91.409(f)(3). Only the items specified in the Work Cto Service. No other representation concerning Airworthiness are expressed or in repair station under work order no. A1137. Authorization Concerning Airworthiness are expressed or in repair station under work order no. A1137. 	 Preserve APU prior to removal – Preserved APU P/N: 3800804-1 S/N: P-292C IAW Honeywell 36-15-49-20-00 Rev 1, dated May 15/114 Removed APU per customer request – Removed APU P/N: 3800804-1 S/N: P-292C from CL604 S/N ACTT: 6104.50 ACTC: 4131. IAW CL604 AMM 49-10-004 I certify the above maintenance/inspection was performed in accordance with current Federal Aviation Regulations. All inspendent in accordance with 91.409(f)(3). Only the items specified in the Work Order have been found to be Airworthy and to Service. No other representation concerning Airworthiness are expressed or implied. Pertinent details of work performed. 				

^{*} Signature and date in this column will be taken as a certificate that all mandatory requirements applicable thereto have been complied with, pertaining to:

(1) Particulars of all maintenance work done on the engine. (2) Particulars of all overhauls, repairs, replacement, modifications and mandatory inspections relating to the engine or Any of its accessories. (3) Particulars of any defects occurring in the engine, and of the rectification of such defects, including a reference to the relevant entries in the technical log.

1.Approving Civil Aviation Authority/Country:	2.				3. Form Tracking Number.
FAA/UNITED STATE	S	AUTHORIZED I	RELEASE CE	RTIFICATE PROVAL TAG	368277
4. Organization Name and Add		17011 0100 017	III (VVOI (TIMI 4E 00 7 ti	THOUSE INC	5. Work Order/Contract/ Invoice Number:
StandardAero Business Aviation Se 1550 Hangar Road Augusta, Georgia 30906 (GW4R22					368277
6. Item: 7. Description:		8. Part Number:	9. Quantity	10. Serial Number:	
1 GTCP36-1	50CL	3800804-1	1	P-292C	REPAIRED
12.Remarks:	, ,		44.		
APU TSN: 6742.0 WARNING: FUEL SYSTEM I NOTE: APU S/N P-292C IS S A SERVICEABLE LCV AND	ED I/A/W GT PRESERVE SHIPPED W LOP SWITC ied in block	CP36-150CL EMM # 4: D. ITHOUT LCV AND LOF CH IS TO BE INSTALLE 11/ 12 was carried out i	9-24-75 REV. 1 / P SWITCH INSTA D BEFORE APU in accordance wi	AND GTCP36-150 ALLED. IS RETURNED T th EASA Part-145	OCL CMM # 49-24-74 REV. 4. O SERVICE. and in respect to that work the aircraft
13a. Certifies the Item identified above	re were manufac	tured in conformity to:	14a. 🛭 14 CFR	43.9 Return to Servi	ce 🗵 Other regulation specified in Block 12
☐ Approved design data and are	In a condition	n for safe operation.	Certifies that u	nless otherwise spe	cified in Block 12, the work identified in Block 11 and
☐ Non-approved design data sp	ecified in Bloc	k 12			plished in accordance with Title 14, Code of Federal o that work, the items are approved for return to service.
13b. Authorized Signature:	13c. Appro	val/Authorization No:	14b. Authorized	Signature:	14c. Approval/Certificate No: CRS# GW4R221M
13d. Name (Type or Printed):	13e. Date (dd/mmm/yyyy):	14d. Name (Typ RONALD		14e. Date (dd/mmm/yyyy): 30/MAR/2022
	The state of the s	User/I	nstaller Respon	sibilities	
Where the user/installer perform	ns work in ac s essential tha	cordance with the nationa at the user/installer ensure	al regulations of an	airworthiness auth	y to install the aircraft engine/propeller/article. ority different than the airworthiness authority of the ccepts aircraft engine(s)/propeller(s)/article(s) from the

Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.

	11) HOURS BROUGHT FORWARD	RAFTType	avy Uana		Mins	
D:	StandardAero	StandardAero 1550 Hangar Roa Augusta Ga. 309 CRS GW4R221M	d		Mins	5) and Da
	C&L AEROSPACE	ENGINE GTCP36-150CL	<i>s/N</i> P-292C	TSN 6742.0	CYCLES N/A	40.00
	REF. W/O # 368277					
	APU Fuel System Component: Differential-Pressure-Ind: Magnetic Chip Detector. C/W Long Term Preservation All Work Accomplished Per CMM # 49-24-74 Rev. 4.	Per CB 40 700-		opection of the AP	er U	
Algoria Septe	WARNING: FUEL SYSTEM PRESERVED. NOTE: APU S/N P-292C IS SHIPPED WI A SERVICEABLE LCV AND LOP SWITCH I	THOUT LCV AND LOP SWI'S TO BE INSTALLED BEFY	FCH INSTALLED.			
April		has been inspect	ed and work		ve	

1. Approving Civil Aviation 3. Form Tracking Number: Authority/Country: AUTHORIZED RELEASE CERTIFICATE FAA/United States 19-014-G011/1 FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG 4. Organization Name and Address: 5. Work Order/Contract/Invoice Number: Aircraft Engine Inspection Inc., 2365 Sunrise Oval, Medina Ohio 44256 CRS# 2AER160B PO# 201902 6. Item: 7. Description: 8. Part Number: 9. Quantity: 10. Serial Number: 11. Status/Work: **APU** P#3800804-1 1 P-292C INSPECTED M#36-150CL 12. Remarks: AEI DID NOT OPEN THIS ENGINE. COMPLIED WITH BORESCOPE REQUEST FOR PO# 201902. BOX 11 WAS INSPECTED IAW ALIED SIGNAL AMM 49-20-00 DATED MAY 16/97. A CURSORY EXTERIOR VISUAL INSPECTION FOR OBVIOUS EXTERNAL LEAKS AND DAMAGE. BORESCOPE INSPECTION OF THE HIGH PRESSURE COMPRESSOR, COMBUSTION CHAMBER, FUEL NOZZLES, AND HPT VANES FOR GENERAL CONDITION AND WEAR. BORESCOPE INSPECTION OF THE VISIBLE HPT 1 AND HPT 2

BLADES. AEI DID NOT CLOSE THIS ENGINE. TSN 6739 CSN: UNK

"This Certificate cancels the disposition in block 12 TSN: 7,160.0 of the Certificate 19-014-G011 dated 30/AUG/2019 and does not cover conformity/condition/release to service"

"CERTIFIES THAT THE WORK SPECIFIED IN BLOCK 12/13 WAS CARRIED OUT IN ACCORDANCE WITH EASA PART-145 AND IN RESPECT TO THAT WORK THE COMPONENT IS CONSIDERED READY FOR RELEASE TO SERVICE UNDER EASA PART-145 APPROVAL NUMBER: "EASA.145.6550"

13a. Certifies the items identified above were	e manufactured in conformity to:	14a. 🛛 14 CFR 43.9 Return to Service	Other regulation specified in Block 12
Approved design data and are in a Non-approved design data specifie		Certifies that unless otherwise specified in Block and described in Block 12 was accomplished in Federal Regulations, part 43 and in respect to the return to service.	ccordance with Title 14, Code of
13b. Authorized Signature:	13c. Approval/Authorization No.:	14b. Authorized Signature:	14c. Approval/Certificate No.: 2AER160B
13d. Name (Typed or Printed):	13e, Date (dd/mmm/yyyy):	14d. Name (Typed or Printed): Jeffrey S. Melvin, Chief Insp.	14e. Date (dd/mmm/yyyy): 27/JAN/2020

User/Installer Responsibilities

It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article,

Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.

Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.

FAA Form 8130–3 (02–14)
NSN: 0052-00-012-9005

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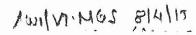
		r						
Auth	roving Civil Aviation ority/Country: //United States	2.	HORIZED RELE	ASE CERT	IFICATE		3. Form Tracking Number: 20150000604698Y15 320322609	
		<u> </u>	m 8130-3, AIRWOR	THINESS APP	PROVAL TA	G		
	¥	Honeywell International Ir 1944 E Sky Harbor Circle PHOENIX AZ 85034	С	Repair Station ZN3R030M			5. Work Order/Co AW-PO-0499-2 320322607 Page 1 of 1	ontract/Invoice Number: 2014-15
	7. Description:		8. Part Number;		9. Quantity:	10. Serial Numbe	ır.	11. Status / Work:
001	ENGINE GAS TURB GTO	CP36-150[CL]	3800804-1		1	P-292C		MODIFIED
CMM 4 MODIS TSN: SEE A HONEY READY F	TTACHED DOCUMENTS WELL CERTIFIES THAT THE WFOR RELEASE TO SERVICE UNI	A.W. SB 49-7885 AS APPLICABLE FOR WORK SPECIFIED IN BLOCK 11/1. DER EASA PART 145 APPROVAL	ORK PERFORMED 2 WAS CARRIED OUT IN ACCORD 3, NO. EASA 145,4136	PANCE WITH EASA PAR	RT 145 AND IN RESI	PECT TO THAT WORK	CTHΕ[PRODUCT/AI	RTICLE) IS CONSIDERED
13a. Ce	difies the items identified about			14a. X14 CFR 43	.9 Return to Servi	ce Sother regulat	ion specified in Bl	ock 12
	Non-approved design data	are in a condition for safe ope specified in Block 12.	ration.	Certifies that unless described in Block 1 Regulations, part 43	2 was accomplish	ied in accordance w	ith Title 14 Code	of Federal
13b. Auf	thorized Signature:	13c. Approva	l/Authorization No.:	14b. Authorized Sign	nature:		14c. Approv	/al/Certificate No.:
				Add	no Pulido		ZN3R030	М
13d. Na	me (Typed or Printed):	13e. Date(dd	/mmm/yyyy):	14d. Name (Typed o	or Printed):		14e. Date(d	d/mmm/yyyy):
	····			Arturo Pulido			27/MAR/20	015

User / Installer Responsibilities

It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article. Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block1. Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.

FAA FORM 8130 - 3 (02-14)

NSN: 0052 - 00 - 012 - 9005



			ULTIN	MATE	LIFE	PART	CARD	
PART NAME		Compress	or Rotor		_	PART NUM	BER	
SERIAL N	UMBER	14-162045-	01841		-	ASSY. PAR	T NUMBER	3822632-1
D	DATE		AIRCRAFT	TIME ON PART THIS INSTALLATION		TOTAL TIME	ON THIS PART	REMARKS
INSTALLED	REMOVED	SERIAL NO.	SERIAL NO.	CYCLE	HOURS	CYCLE	HOURS	HONEYWELL
Mar-15		15P12756		00:00	00:00	00:00	00:00	P-191C ZN3R030M
Ho	ne	yw	ell		<u>L</u>	Engines & Sp.O. Box 52		

			ULTIN	MATE	LIFE	PART	CARD			
PART NAM	ΜE	Turbine Rotor				PART NUM	BER			
SERIAL N	UMBER	14-156101-	04782		ASSY, PART NUMBER			38423		
DATE		ENGINE	AIRCRAFT		TIME ON PART THIS INSTALLATION		TOTAL TIME ON THIS PART		REMARKS	
INSTALLED	REMOVED	SERIAL NO.	SERIAL NO.	CYCLE	HOURS	CYCLE	HOURS		HONEYWELL	
Mar-15		15P12756	1:	00:00	00:00	00:00	00:00	P-292C		
	2									
10										
2 (8) 10 2,103										
Ho	ne	yw	ell			Honeywell Ir Engines & Sy P.O. Box 521 Phoenix, Ari	ystems 181		A	X6167-2B